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Structure, mechanical and thermal properties of Y-doped CrAlN coatings

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Abstract: CrAlYN coatings with different Y contents (0, 5 and 12 at.%) were deposited by cathodic arc evaporation to investigate the influence of Y-addition on the structure, mechanical and thermal properties of CrAlN coatings by using X-ray diffraction, scanning electron microscopy, differential scanning calorimetry, thermal gravimetric analysis and nanoindentation. The structural transformation of single phase cubic $Cr_{0.42}Al_{0.58}N$ and $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coatings to cubic–wurtzite mixed $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coating leads to a drop in hardness from (30.2±0.7) GPa of $Cr_{0.42}Al_{0.58}N$ and (32.0±1.0) GPa of $Cr_{0.39}Al_{0.56}Y_{0.05}N$ to (25.2±0.7) GPa of $Cr_{0.32}Al_{0.56}Y_{0.12}N$. The incorporation of 5 at.% Y retards the thermal decomposition of CrAlN, verified by the postponed precipitation of *w*-AlN and N-loss upon annealing. Correspondingly, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coating consistently exhibits the highest hardness value during thermal annealing. Nevertheless, alloying with Y exerts an adverse effect on the oxidation resistance of CrAlN.

Key words: CrAlYN coatings; cathodic arc evaporation; structural evolution; hardness; thermal stability; oxidation resistance

1 Introduction

Transition metal nitrides have been widely used as protective coatings for tools, dies and components owing to their high hardness, superior excellent oxidation resistance, tribological properties and promising corrosion resistance [1-4]. Amongst them, CrAlN coatings deposited by physical vapor deposition (PVD) have gained extensive attention as a promising type of wear-resistant hard coatings. CrAlN coatings are developed by incorporating Al atoms into CrN lattices, replacing a part of the Cr atoms to form a metastable substitutional solid solution with superior mechanical properties [5]. Besides, CrAlN coatings reveal exceptional oxidation resistance as a dense and protective (Al,Cr)₂O₃ oxide layer is formed in the early stage of oxidation inhibiting O^{2-} from diffusing inward [6]. Nevertheless, the

high-temperature applications of CrAlN coatings are still limited due to their relatively inferior thermal stability, for which their hardness reveals a dramatic downtrend after the temperature exceeding 900 °C. Under exposure to high temperatures, CrAlN coatings undergo thermal decomposition in two forms. For one thing, the metastable *c*-AlN transforms into its stable phase wurtzite (*w*-) AlN with low hardness. For another, the unstable Cr—N bonds break and Cr with body-centered cubic (bcc) structure is formed via an intermediate phase hexagonal (*h*-) Cr₂N [7].

To further improve the comprehensive properties of CrAlN coatings, alloying with the fourth element (V [8], Ta [9], Zr [10], Si [11] and B [12], etc) has been developed as an effective method due to its potential to retard the diffusioninvolved processes like recovery, decomposition and recrystallization. Among them, V-addition effectively enhances the thermal stability of CrAlN

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coatings by suppressing the formation of w-AlN and the dissociation of Cr-N bonds [13]. Besides, V-addition reduces the friction coefficient at high temperatures on account of the formation of lubricious oxides in CrAlVN coatings [14]. Adding moderate Ta is found to improve the thermal stability and mechanical properties of CrAlN coatings [9]. Zr-alloyed CrAlN coatings show excellent high temperature tribological properties for the formation of a Cr₂O₃-rich layer that reduces friction coefficient and wear rate [15]. Si-doped CrAlN coatings with nanocomposite structure obtain not only superior oxidation resistance but also improved corrosion resistance [16,17]. Alloying with B brings about a significant enhancement in mechanical properties and oxidation resistance compared with CrAlN coatings [18].

Apart from the mentioned alloying elements above, Y-incorporation has gained great attention in recent years as it significantly enhances the oxidation resistance of CrAlN coatings [19,20]. The segregation of yttrium atoms at grain boundaries and the formation of Al-rich oxide layers block the diffusion of cations (especially Cr ions) and anions, thus perfecting the oxidation resistance of CrAlN coatings [20]. There are also some reports on the mechanical properties [21] and thermal stability [22] of CrAlYN coatings. The hardness of CrAlN coatings is increased by alloying with a small amount of Y owing to solid solution mechanism and Hall-Petch effect [20]. In terms of thermal stability, incorporation of yttrium into CrAlN coating effectively retards the diffusion-associated processes [22]. Hence, Y-addition at a suitable amount is a feasible approach to improve the hardness, oxidation resistance and thermal stability of CrAlN coatings.

At present, the studies are mainly focused on CrAlYN coatings with a low content of yttrium (around 1 at.%) [19,20,23,24], while there are less systematic studies involving the performances of CrAlYN coatings with higher yttrium contents. In addition, the detailed phase evolution during the oxidation process of CrAlYN coatings, which is vital to ascertain the mechanism of Y tailoring the oxidation resistance of CrAlN coatings, has not been investigated before. Therefore, CrAlYN coatings with a wide range of Y contents (0, 5 and 12 at.%) were prepared in this work to investigate their structure, mechanical properties, oxidation resistance and thermal stability.

2 Experimental

2.1 Coating depositions

CrAlN and CrAlYN coatings were deposited on various substrates in a set of commercial equipment (Oerlikon Balzers RCS) by cathodic arc evaporation using $Cr_{0.40}Al_{0.60}$, $Cr_{0.35}Al_{0.60}Y_{0.05}$, and Cr_{0.30}Al_{0.60}Y_{0.10} (99.99% in purity) targets, respectively. The used substrates include cemented carbides (WC-6wt.%Co), steel foils, tungsten (W) plates, and polycrystalline Al₂O₃ sheets, varying with the determination methods. To enhance the adhesion between substrate and coating, the substrates were ultrasonically cleaned in ethanol. Before deposition, the chamber was vacuumed to a pressure lower than 1.0×10^{-3} Pa and heated to 550 °C. Subsequently, an argon discharging process was conducted for 30 min to clean the substrates with a target-substrate distance of 240 mm. Afterward, coatings were deposited under a nitrogen (N₂, 99.99% in purity) atmosphere at approximately 3.0 Pa with a target current of 140 A and a bias voltage of -40 V.

2.2 Post-deposition treatments

Prior to thermal treatments, the coated lowalloying steel foils were dissolved in nitric acid to separate substrates and coatings. Then, the obtained flakelets of coatings were ground into powder. To study the thermal stability of coatings, the obtained powder samples were placed in a differential scanning calorimetry (DSC) instrument (Netzch STA 409C) under a pure Ar atmosphere and gradually heated up from room temperature (RT) to a specific annealing temperature ($T_a=900$, 1000, 1100, 1200, 1300, 1450 and 1550 °C) with a heating rate of 10 °C/min. Afterward, they were cooled down at a rate of 50 K/min. Besides, the samples onto W plates were placed in a vacuum furnace (COD533R, pressure $\leq 10^{-3}$ Pa) for vacuum annealing. The furnace was heated to a specific temperature (T_a =800–1100 °C) with a heating rate of 10 °C/min for heat preservation of 30 min. Finally, the samples were naturally cooled down to RT with furnace cooling.

Synchronous DSC and thermal gravimetric analysis (TGA) experiments in flowing synthetic air

(79 vol.% N₂, 21 vol.% O₂, 20 mL/min) were performed to measure the heat flow and mass change on the same thermal analysis instrument from RT to 1500 °C with a heating rate of 10 °C/min and a cooling rate of 50 °C/min. The isolation oxidation experiments of coated polycrystalline Al₂O₃ substrates were performed in a DSC instrument at 800 °C for 15 h in synthetic air.

2.3 Characterizations

The surface and fracture cross-sectional morphologies of as-deposited and oxidized samples were investigated by scanning electron microscope (SEM, Zeiss Supra 55). The compositions of as-deposited coatings were determined by an energy-dispersive X-ray spectroscopy (EDX, Oxford Instrument X-MaX) detector attached to SEM. The crystal structure of specimens in various states was examined by X-ray diffraction (XRD), using a Bruker D8 Advance diffractometer in the Bragg-Brentano configuration with Cu K_{α} source $(\lambda = 1.541 \text{ Å})$. Hardness (H) and elastic modulus (E) measurements of coated W specimens were carried out with a nanoindentation instrument (Anton Paar UNHT³) with a Berkovich diamond tip following the Oliver-Pharr method [25]. A penetration load of 15 mN can ensure that the indentation depth (~160 nm) is less than 10% of the coating thickness, avoiding the interference of substrates. Average values and standard deviations were obtained from at least 20 indents.

3 Results and discussion

3.1 Chemical composition, structure, and morphology

Based on EDX measurements, the prepared CrAlN and CrAlYN coatings are stoichiometric as they present a nitrogen content of ~50 at.%. Their nominal compositions are $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}$ - $Y_{0.05}N$ and $Cr_{0.32}Al_{0.56}Y_{0.12}N$, corresponding to $Cr_{0.40}Al_{0.60}$, $Cr_{0.35}Al_{0.60}Y_{0.05}$ and $Cr_{0.30}Al_{0.60}Y_{0.10}$ targets, respectively. The composition deviations between coatings and the counterpart targets are ascribed to the different atomic evaporation rates of each element, the scattering and re-sputtering process [26]. Figure 1 presents the XRD patterns of as-deposited coatings. $Cr_{0.42}Al_{0.58}N$ and $Cr_{0.39}Al_{0.56}$ - $Y_{0.05}N$ coatings present a single phase face-centered cubic (fcc) structure. $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coating

exhibits a mixed structure of cubic and wurtzite, as better observed in the XRD pattern of fine scanning in the inset of Fig. 1, indicating that Y-incorporation decreases the metastable solubility limitation of c-AlN in Cr—N. Atomic radius of Y (1.80 Å) [27] is larger than that of Cr (1.40 Å) [27] or Al (1.25 Å) [27], leading to the expansion of CrAlN lattice and an increase of lattice parameter [28] from (4.1072±0.0015) Å $(Cr_{0.42}Al_{0.58}N)$ to (4.1594 ± 0.0009) Å (Cr_{0.39}Al_{0.56}Y_{0.05}N), and then to (4.1938 ± 0.0105) Å $(Cr_{0.32}Al_{0.56}Y_{0.12}N)$. The XRD peaks shift to lower 2θ diffraction angles with increasing Y content, further verifying the lattice expansion by Bragg's law. Besides, alloying with 12 at.% Y causes an obvious widening of the diffraction peaks, indicating the decreased grain size of Cr_{0.32}Al_{0.56}Y_{0.12}N coating. The mean crystallite sizes of $\sim 327 \text{ nm}$ for $Cr_{0.42}Al_{0.58}N$, ${\sim}134~nm$ for $Cr_{0.39}Al_{0.56}Y_{0.05}N$ and ${\sim}5~nm$ for Cr_{0.32}Al_{0.56}Y_{0.12}N are roughly calculated using the Scherrer equation [29].



Fig. 1 XRD patterns of as-deposited $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coatings

Figure 2 displays SEM surface and fracture cross-sectional morphologies of CrAlN and CrAlYN coatings. All coatings present defects derived from the evaporation of target macroparticles with different melting points during evaporation process. PANJAN et al [30] reported that cathode materials with a low melting point can give rise to larger droplets, which explains the phenomenon that the droplet size of Y-alloying coatings is increased. The fracture cross-sections convey that the thickness of all coatings is \sim 3.0 µm. Cr_{0.42}Al_{0.58}N (Fig. 2(d)) exhibits a typically dense



Fig. 2 SEM surface (a–c) and fracture cross-sectional (d–f) morphologies of as-deposited $Cr_{0.42}Al_{0.58}N$ (a, d), $Cr_{0.39}Al_{0.56}Y_{0.05}N$ (b, e), and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ (c, f) coatings

and columnar crystal morphology. Doping with Y hinders the growth of the columnar crystal and leads to the grain refinement. Cr_{0.39}Al_{0.56}Y_{0.05}N (Fig. 2(e)) shows a fine columnar structure. As Y content increases to 12 at.%, obvious grain refinement and nanocrystalline morphology of $Cr_{0.32}Al_{0.56}Y_{0.12}N$ are observed, as shown in Fig. 2(f). The grain refinement of Cr_{0.39}Al_{0.56}Y_{0.05}N coating stems from the strain induced by lattice expansion. And nanocrystalline the morphology of Cr_{0.32}Al_{0.56}Y_{0.12}N coating is related to the competitive growth of cubic and wurtzite phases, which dramatically constrains the growth of columnar crystal. Moreover, the formation of wurtzite phase provides plenty of nucleation sites [31], which also contributes to the formation of nanocrystalline morphology.

3.2 Mechanical properties

Figure 3 depicts the nanoindentation hardness and elastic modulus of CrAlN and CrAlYN coatings. The hardness of $Cr_{0.39}Al_{0.56}Y_{0.05}N$ rises to (32.0±1.0) GPa due to the solution strengthening and Hall–Petch effect (grain refinement) [20,32], compared to a hardness of (30.2±0.7) GPa of the



Fig. 3 Nanoindentation hardness and elastic modulus of as-deposited $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}Y_{0.05}N$, and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coatings

 $Cr_{0.42}Al_{0.58}N$. However, the hardness of $Cr_{0.32}Al_{0.56}$ -Y_{0.12}N coating decreases to (25.2±0.7) GPa due to the formation of *w*-AlN. As mentioned above, doping with yttrium gives rise to the decrease of grain size. KIM and BUSH [33] reported that grain refinement could contribute to a reduction of elastic modulus. Thus, the elastic modulus decreases from (488.2±23.6) GPa of $Cr_{0.42}Al_{0.58}N$ to (432.4±14.2) GPa of $Cr_{0.39}Al_{0.56}Y_{0.05}N$. In addition, the formation of *w*-AlN leads to a further deduction in elastic modulus of $Cr_{0.32}Al_{0.56}Y_{0.12}N$ to (306.0±7.6) GPa with the combined effect of grain refinement [33].

3.3 Thermal stability

To investigate the impact of Y-addition on the structural evolution of CrAlN coatings during annealing, XRD determinations were performed for Cr_{0.42}Al_{0.58}N, Cr_{0.39}Al_{0.56}Y_{0.05}N, and Cr_{0.32}Al_{0.56}- $Y_{0.12}N$ coatings annealed at given temperatures, as shown in Fig. 4. After annealing at 900 °C, $Cr_{0.42}Al_{0.58}N$ coating maintains original its single-phase fcc structure with a slight shift of peaks to a higher diffraction angle, which is associated with the recovery process involving stress relaxation and the annihilation of defects [34]. As the annealing temperature (T_a) increases to 1000 °C, a weak diffraction peak of w-AlN at $2\theta = -33.2^{\circ}$ is observed. After annealing at $T_a=1100$ °C, *h*-Cr₂N, which is the product of Cr—N dissociation [35], is detected at $2\theta = -37.3^{\circ}$, 40.2° and 42.6°. With increasing T_a to 1200 °C, the continuous thermal decomposition leads to elevated diffraction intensities of w-AlN and h-Cr₂N. And bcc-Cr is formed due to the further dissociation of Cr-N bonds of h-Cr2N at 1200 °C. Besides, the original peaks of c-(Cr,Al)N can hardly be detected. At 1450 °C, the N-loss process terminates and the h-Cr₂N has fully transformed into bcc-Cr. Finally, Cr_{0.42}Al_{0.58}N completely decomposes into w-AlN and bcc-Cr at 1550 °C.

On the other side, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coating (Fig. 4(b)) maintains its single phase structure at 900 and 1000 °C. The introduction of 5 at.% Y retards the precipitation of w-AlN by 100 °C in comparison with Cr_{0.42}Al_{0.58}N. h-Cr₂N exhibits lower XRD diffraction intensity than that in Cr_{0.42}Al_{0.58}N after annealing at 1100 °C. At 1200 °C, bcc-Cr and fcc-YN are observed, while the original diffraction peaks of cubic phase disappear. Further increasing T_a to 1300 °C, the intensity of bcc-Cr peak becomes more pronounced with the expense of h-Cr₂N. Finally, after annealing at 1450 and 1550 °C, Cr_{0.39}Al_{0.56}Y_{0.05}N completely decomposes into w-AlN, fcc-YN and bcc-Cr. For Cr_{0.32}Al_{0.56}- $Y_{0.12}N$ coating, w-AlN is formed during deposition, which also exists with elevating T_a from 900 to 1450 °C (Fig. 4(c)). Likewise, h-Cr₂N is generated after annealing at $T_a=1100$ °C, with a lower intensity than that in $Cr_{0.42}Al_{0.58}N$. The generating temperature of fcc-YN is advanced to 1100 °C due to the increase of Y content. According to previous research, the bonding energy of Y-N (477 kJ/mol) is higher than that of Cr-N (378 kJ/mol) and Al - N (368 kJ/mol) [36]. Besides, ab initio calculations with experiment investigated by ROVERE et al [37] also verified that fcc-YN reveals higher stability than c-CrN. Hence, doping with Y effectively stabilizes the N atoms for the Y-containing coatings, especially Cr_{0.32}Al_{0.56}Y_{0.12}N. With increasing T_a to 1300 °C, the diffraction intensities of w-AlN, h-Cr₂N and bcc-Cr continuously increase, suggesting an aggravated



Fig. 4 XRD patterns of $Cr_{0.42}Al_{0.58}N$ (a), $Cr_{0.39}Al_{0.56}Y_{0.05}N$ (b), and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ (c) coatings after annealing up to 1550 °C

thermal decomposition process. After annealing at 1450 °C, the diffraction peaks of *h*-Cr₂N in Cr_{0.32}Al_{0.56}Y_{0.12}N are still prominent, while those in Cr_{0.42}Al_{0.58}N and Cr_{0.39}Al_{0.56}Y_{0.05}N disappear due to the intensified dissociation of Cr—N bonds. Finally, Cr_{0.32}Al_{0.56}Y_{0.12}N coating is composed of *w*-AlN, fcc-YN, *h*-Cr₂N and bcc-Cr at T_a =1550 °C.

The hardness values of coatings during annealing strongly depend on the above-mentioned phase evolution. Figure 5 illustrates the hardness of CrAIN and CrAIYN coatings as a function of annealing temperature. At 900 °C, the annihilation of defects and the relaxation of residual stress lead to a slight reduction in hardness of Cr_{0.42}Al_{0.58}N from (30.2±0.7) GPa in as-deposited state to (28.8 ± 0.9) GPa. The precipitation of w-AlN induces a continuous decline of hardness to (27.6 ± 0.8) GPa after annealing at 1000 °C. After annealing at 1100 °C, the hardness of $Cr_{0.42}Al_{0.58}N$ shows a significant decrease to (22.6 \pm 0.7) GPa as *h*-Cr₂N is formed and N-loss becomes severe. No distinct variation in the hardness of $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coating is obtained till increasing T_a to 1000 °C. Even after annealing at 1100 °C, Cr_{0.39}Al_{0.56}Y_{0.05}N coating can still maintain a hardness value higher than 31 GPa. $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coating reveals an obvious enhanced ability to maintain its mechanical properties at high temperatures, further verifying the beneficial effect of Y-addition with an amount of 5 at.% on thermal stability. For Cr_{0.32}Al_{0.56}Y_{0.12}N coating, the hardness reaches its peak value of (28.3 ± 0.5) GPa at 900 C. This can be attributed to a more compact microstructure induced by the



Fig. 5 Hardness of $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}Y_{0.05}N$, and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coatings as function of annealing temperature T_a

intensified diffusion process [38], which effectively counteracts negative effects of w-AlN formation and recovery process. The hardness of maintain stable $Cr_{0.32}Al_{0.56}Y_{0.12}N$ can after annealing at 1000 °C, owing to the postponed dissociation of Cr-N bonds. However, annealing at 1100 °C leads to a significant drop of hardness to (21.5 ± 0.5) GPa because of the precipitation of h-Cr₂N and N-loss.

3.4 Oxidation resistance

Figure 6 presents the synchronous TGA and DSC curves from 500 to 1500 °C of Cr_{0.42}Al_{0.58}N, $Cr_{0.39}Al_{0.56}Y_{0.05}N$, and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coatings in synthetic air. Y-incorporation results in an obvious drop in the onset temperature of oxidation, which can be identified by both the starting temperature of exothermic features on DSC curve (Fig. 6(a)) and the initial temperature of mass gain on TGA curve (Fig. 6(b)). By analyzing the data of heat flow, the oxidation process of Cr_{0.42}Al_{0.58}N starts at ~1165 °C, while Cr_{0.39}Al_{0.56}Y_{0.05}N initiates an exothermic feature of oxidation at ~1055 °C. Increasing the Y content to 12 at.% leads to a lower onset temperature of ~780 °C. The TGA curves in Fig. 6(b) also reveal that the onset mass gain temperature of Y-containing coatings is lower than



Fig. 6 Simultaneous thermal analysis in synthetic air of $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ coatings with temperature up to 1500 °C: (a) TGA; (b) DSC

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that of $Cr_{0.42}Al_{0.58}N$, further verifying the negative effect of Y-incorporation on oxidation resistance. It worth noting that $Cr_{0.39}Al_{0.56}Y_{0.05}N$ and is Cr_{0.32}Al_{0.56}Y_{0.12}N coatings both exhibit an extra endotherm feature initiated at ~1180 and ~1090 °C, respectively. This can be ascribed to the dissociation process of Cr-N bonds in CrAlYN coatings, which is endotherm in nature [7]. Besides, the TGA curves of Cr_{0.39}Al_{0.56}Y_{0.05}N and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ begin to show a downtrend at ~1180 and ~1090 °C, respectively, which is consistent with the DSC analysis as the breaking of Cr-N bonds is accompanied with a mass loss [9]. In conclusion, DSC and TGA analyses indicate that alloying with Y exerts an adverse impact on the oxidation resistance of CrAlN coatings.

Figure 7 presents the XRD patterns of $Cr_{0.42}Al_{0.58}N$, $Cr_{0.39}Al_{0.56}Y_{0.05}N$, and $Cr_{0.32}Al_{0.56}$ after $Y_{0.12}N$ coatings oxidation at given temperatures. $Cr_{0.42}Al_{0.58}N$ coating can keep unoxidized at 1000 °C, as shown in Fig. 7(a). After oxidation at 1100 °C, the diffraction peaks of α -Cr₂O₃ (space group $R\overline{3}c$) and/or α -Al₂O₃ (space group R3c) occur, which are more prominent with elevating $T_{\rm a}$ to 1200 °C. Increasing $T_{\rm a}$ to 1300 °C leads to the breaking of Cr-N bonds and N-loss, which is demonstrated by the emergence of h-Cr₂N. After the oxidation at 1400 °C, the original diffraction peaks of cubic phase disappear, and Cr_{0.42}Al_{0.58}N coating is completely oxidized into and their mixed α -Cr₂O₃, α -Al₂O₃ phase α -(Cr,Al)₂O₃ (space group $R\overline{3}c$).

For $Cr_{0.39}Al_{0.56}Y_{0.05}N$, α - Cr_2O_3 and α - Al_2O_3 are detected at 1000 °C, indicating that Y-incorporation decreases the onset temperature of oxidation. At 1100 °C, feeble peaks of orthometric structure (o-) YCrO₃ (space group *Pnma*) and *o*-YAlO₃ (YAP, space group *Pnma*) are observed. With increasing oxidation temperature to 1200 °C, the original diffraction signals of cubic phase are nearly invisible in $Cr_{0.39}Al_{0.56}Y_{0.05}N$, further manifesting a weakened oxidation resistance. The formation of N-loss h-Cr₂N indicates the process in Cr_{0.39}Al_{0.56}Y_{0.05}N, corresponding to its TGA curve in Fig. 6(b) which starts to go down at 1200 °C. In addition, a small amount of cubic (c-) Y₂O₃ (space group $1\overline{a}3$) are detectable in Cr_{0.39}Al_{0.56}Y_{0.05}N after being oxidated at 1200 °C. By further increasing the temperature to 1300 °C, c-Y₂O₃ is not detected as it combines with Cr and/or Al to form the orthometric YCrO₃ and/or YAlO₃, which mainly aggregates at the grain boundaries for Y-doped coatings [39,40]. Finally, Cr_{0.39}Al_{0.56}Y_{0.05}N is composed of α -(Cr,Al)₂O₃, *o*-YCrO₃, *o*-YAlO₃ and *h*-Cr₂N at 1400 °C. The phase evolution of Cr_{0.32}Al_{0.56}Y_{0.12}N is similar to that of Cr_{0.39}Al_{0.56}Y_{0.05}N except for the existing temperature of c-Y2O3 as well as the formation of h-YAlO₃ and c-Y₃Al₅O₁₂. c-Y₂O₃ can be detected at 1200 and 1300 °C as a higher Y content elevates the temperature where it is fully consumed. Additionally, another YAlO₃, designated yttrium aluminum hexagonal (YAH), with space



Fig. 7 XRD patterns of $Cr_{0.42}Al_{0.58}N$ (a), $Cr_{0.39}Al_{0.56}Y_{0.05}N$ (b), and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ (c) powdered samples after oxidation in synthetic air up to 1400 °C



Fig. 8 SEM fracture cross-sectional morphologies and EDX line-scan profiles of $Cr_{0.39}Al_{0.56}Y_{0.05}N$ (a) and $Cr_{0.32}Al_{0.56}-Y_{0.12}N$ (b) coatings on polycrystalline Al_2O_3 substrates after isothermal oxidation in synthetic air at 800 °C for 15 h

group of $P6_3/mmc$, occurs at 1100 °C. After oxidation at 1200 °C, $Y_3Al_5O_{12}$ (YAG), with a garnet structure of cubic system (space group $1\overline{a}3d$), emerges with an obvious diffraction peak at ~35.5°, which disappears at 1400 °C. Furthermore, *h*-YAlO₃ gradually transforms into *c*-Y₃Al₅O₁₂ at 1200 °C, which is also not detectable at 1400 °C [41].

Figure 8 depicts the cross-sectional SEM morphologies and EDX line-scan profiles of Cr_{0.39}Al_{0.56}Y_{0.05}N and Cr_{0.32}Al_{0.56}Y_{0.12}N coatings after isothermal oxidation at 800 °C for 15 h. Cr_{0.42}Al_{0.58}N coating possesses excellent oxidation resistance so that it is not affected after oxidation at 800 °C for 15 h according to the previous report [6]. By contrast, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ (Fig. 8(a)) and Cr_{0.32}Al_{0.56}Y_{0.12}N (Fig. 8(b)) reveal thick oxide scales of \sim 3.9 and \sim 4.5 µm, respectively, suggesting that Y-addition reduces the oxidation resistance of CrAlN coatings. EDX line-scan analysis was performed on the oxidized CrAlYN samples to further explore the elemental distribution of the oxide layers. It can be found that the Cr, Al, O, and Y elements are evenly distributed in the oxidized region. The early formation of Y-containing phases including YCrO₃, YAlO₃ and Y₂O₃ located in grain boundaries provides a fast channel for the inward diffusion of oxygen anions [20], and thus weakens the oxidation resistance of Y-containing coatings. Besides, the oxide layers of Cr_{0.39}Al_{0.56}- $Y_{0.05}N$ and $Cr_{0.32}Al_{0.56}Y_{0.12}N$ show a porous structure. Increased thickness of oxide layer with increasing Y-content can be observed by the SEM fracture cross-sectional images and EDX line-scan profiles.

4 Conclusions

(1) $Cr_{0.42}Al_{0.58}N$ and $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coatings both reveal a single phase face-centered cubic structure, but $Cr_{0.32}Al_{0.56}Y_{0.12}N$ presents а cubic-wurtzite mixed structure. The hardness of $Cr_{0.39}Al_{0.56}Y_{0.05}N$ rises to (32.0±1.0) GPa from (30.2 ± 0.7) GPa of Cr_{0.42}Al_{0.58}N owing to the solid solution mechanism and Hall-Petch effect. However, the formation of w-AlN in Cr_{0.32}Al_{0.56}Y_{0.12}N leads to a drop in hardness to (25.2±0.7) GPa.

(2) The addition of 5 at.% Y into CrAlN postpones the formation of *w*-AlN by 100 °C compared with $Cr_{0.42}Al_{0.58}N$. In addition, $Cr_{0.39}Al_{0.56}Y_{0.05}N$ coating consistently exhibits the highest hardness value during annealing.

(3) The introduction of Y results in thicker oxide layers of ~3.9 μ m for Cr_{0.39}Al_{0.56}Y_{0.05}N and ~4.5 μ m for Cr_{0.32}Al_{0.56}Y_{0.12}N after isothermal oxidation at 800 °C for 15 h. Y dramatically weakens the oxidation resistance of CrAlN coating owing to the earlier formation of Y-containing oxide phases (YCrO₃, YAlO₃ and Y₂O₃) which provides a fast channel for oxygen and then the porous and nonprotective oxide layer is formed.

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Y 掺杂 CrAIN 涂层的结构、力学与热性能

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摘 要:采用阴极弧蒸发法制备不同 Y 含量(0、5%和 12%,摩尔分数)的 CrAlYN 涂层,使用 X 射线衍射、扫描 电子显微镜、差示扫描量热仪、热重分析仪和纳米压痕技术研究 Y 掺杂对 CrAlN 涂层结构、力学性能以及热性 能的影响。Cr_{0.42}Al_{0.58}N 和 Cr_{0.39}Al_{0.56}Y_{0.05}N 涂层为单相立方结构,而 Cr_{0.32}Al_{0.56}Y_{0.12} 涂层为立方-六方混合结构, 由于结构的转变,涂层硬度值由 Cr_{0.42}Al_{0.58}N 的(30.2±0.7) GPa 和 Cr_{0.39}Al_{0.56}Y_{0.05}N 的(32.0±1.0) GPa 下降到 Cr_{0.32}Al_{0.56}Y_{0.12}N 的(25.2±0.7) GPa。热退火过程中六方 AlN 相的析出延迟以及 N 损的延迟证明添加 5% Y 能够延 缓 CrAlN 涂层的热分解,相应地,Cr_{0.39}Al_{0.56}Y_{0.05}N 涂层在退火过程中始终表现出最高的硬度值。然而,Y 的掺 杂对 CrAlN 涂层的抗氧化性产生不利的影响。

关键词: CrAlYN 涂层; 阴极弧蒸发; 结构演变; 硬度; 热稳定性; 抗氧化性