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Anti-oxidation properties of ZrB₂ modified silicon-based multilayer coating for carbon/carbon composites at high temperatures

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Abstract: To improve the anti-oxidation ability of silicon-based coating for carbon/carbon (C/C) composites at high temperatures, a ZrB_2 modified silicon-based multilayer oxidation protective coating was prepared by pack cementation. The phase composition, microstructure and oxidation resistance at 1773, 1873 and 1953 K in air were investigated. The prepared coating exhibits dense structure and good oxidation protective ability. Due to the formation of stable $ZrSiO_4$ -SiO₂ compound, the coating can effectively protect C/C composites from oxidation at 1773 K for more than 550 h. The anti-oxidation performance decreases with the increase of oxidation temperature. The mass loss of coated sample is 2.44% after oxidation at 1953 K for 50 h, which is attributed to the decomposition of $ZrSiO_4$ and the volatilization of SiO₂ protection layer.

Key words: C/C composites; coating; ZrB2; anti-oxidation properties

1 Introduction

Carbon/carbon (C/C) composites are attractive materials for thermo-structural components in engines and aerospace vehicles [1,2]. However, the oxidation problem at temperatures above 773 K limits their application in oxygen-containing atmosphere [3,4].

Silicon-based ceramic coatings, especially MoSi₂-SiC coating, have been widely used for oxidation protection for C/C composites at high temperatures [5-7]. The formation of SiO₂ glassy film on the coating surface has very low oxygen permeability (10^{-13} g/(cm·s) at 1473 K and 10^{-11} g/(cm·s) at 2473 K) [8], which can prevent oxygen from diffusing into C/C composites [9,10]. However, with oxidation time increasing or oxidation temperature rising, the MoSi₂-SiC coating exhibits poor oxidation resistance because of the formation of the holes and bubbles in SiO₂ glassy film. Moreover, the unitary SiO₂ glass would volatilize slowly, resulting in the consumption of coating materials [11–14]. Therefore, the formation of a stable vitreous film without holes and bubbles throughout the oxidation process is essential. To solve this problem, ZrB₂ ceramic is introduced to silicon-based coating.

ZrB₂ has an extremely high melting point (3313 K), low coefficient of thermal expansion $(5.5 \times 10^{-6} \text{ K}^{-1})$, high thermal shock resistance, retention of strength at elevated temperatures and stability in extreme environment [15]. ZrB₂-based ceramics are considered the potential candidate materials used in hypersonic vehicles and reusable atmospheric re-entry vehicle [16,17]. ZrO₂ can be formed for the oxidation of ZrB₂ in air at high temperatures. It reacts with SiO₂ to form thermally stable ZrSiO₄ during the oxidation of silicon-based coating [18–20]. Adding ZrB₂ to MoSi₂–SiC coating can be expected to improve the stability of SiO₂ glass and improve the service time of MoSi₂–SiC coating at high temperatures.

Pack cementation method is an effective technique to prepare dense silicon-based anti-oxidation coatings, which can provide strong interfacial bonding in a simple and rapid way [21,22]. SiC ceramic is widely used as an internal buffer layer because of its good anti-oxidation property and compatibility with C/C composites. In addition, to solve the mismatch of thermal expansion coefficient between SiC inner layer and ZrB_2 modified $MoSi_2$ -SiC layer, a ZrB_2 modified SiC-Si coating is

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designed as middle layer, which is expected to decrease the formation of penetrable cracks in coating. A scheme of multilayer coating structure is designed as shown in Fig. 1.



Fig. 1 Designed schematic structure of silicon-based multilayer coating for C/C composites

The aim of this work is to develop a ZrB_2 modified silicon-based multilayer coating, which is expected to improve the oxidation resistance of silicon-based coating for C/C composites. The crystalline structure and morphology of the coating were studied. The oxidation resistance performances at 1773, 1873 and 1953 K were investigated. The oxidation mechanism of prepared coating was also discussed.

2 Experimental

Two dimensional C/C composites with density of 1.70 g/cm^3 were used as substrates. The composites were mechanically abraded using 320 and 800 grit SiC papers, then cleaned with distilled water and dried at 373 K. The first step was to prepare SiC bonding layer on C/C composites by pack cementation with Si, graphite and Al₂O₃ powders at 1973–2273 K for 2 h in argon protective atmosphere [11]. The SiC–Si–ZrB₂ middle layer was prepared at 2073–2373 K for 2 h in argon protective atmosphere, using Si, B₂O₃, graphite and ZrB₂. The preparation details were reported in Ref. [23]. The outer coating was prepared at 2173–2473 K for 2 h with the powder composition of 40%–65% Si, 20%–35% MoSi₂, 8%–15% graphite and 20%–30% ZrB₂ in mass fraction.

A MoSi₂ resistance heated furnace was used for isothermal oxidation test. The samples were exposed to air at 1773, 1873 and 1953 K. During oxidation, the samples were weighed at room temperature by an electronic precision balance with a sensitivity of ± 0.1 mg. Mass loss $\Delta m\%$ was calculated by the following equation:

$$\Delta m = [(m_0 - m_1) / m_0] \times 100\% \tag{1}$$

where m_0 and m_1 are the mass of coated samples before and after oxidation, respectively.

The crystalline structure and morphology of the as-obtained multilayer coating and oxidized coating were

characterized using X-ray diffraction (XRD, X' Pert PRO) and scanning electron microscopy (SEM, JSM-6430) equipped with energy dispersive spectroscopy (EDS).

3 Results and discussion

3.1 Microstructure of multilayer coating

The XRD pattern of obtained coating surface by pack cementation (Fig. 2(a)) shows that the coating is composed of SiC, Si, $MoSi_2$ and ZrB_2 . The backscattering electron image of coating surface (Fig. 2(b)) indicates the formation of a dense structure. The prepared coating has three kinds of crystalline particles. The XRD and EDS analyses show that the white phase is a mixture of $MoSi_2$ and ZrB_2 . The gray black and gray phases are SiC and Si, respectively.



Fig. 2 XRD pattern (a) and SEM image (b) of multilayer coating obtained by pack cementation

Figure 3 shows the cross-section SEM image of the as-received multilayer coating. The coating is about 120 μ m in thickness. No penetration cracks or large holes can be observed in the coating. In addition, no obvious interface can be found between layers, which illustrates the good bonding. Therefore, the dense ZrB₂ modified silicon-based multilayer coating is obtained by pack cementation.



Fig. 3 Cross-section SEM image of coated C/C composites

3.2 Oxidation resistance of multilayer coating at 1773 K

Figure 4 illustrates the isothermal oxidation curves of coated C/C composites at 1773 K. The SiC/MoSi₂– SiC–Si coating provides 150 h oxidation protection for C/C composites at 1773 K. After the introduction of ZrB₂ to silicon-based coating, the coated multilayer coating shows better oxidation resistance ability. It can efficiently protect C/C composites from oxidation at 1773 K for more than 550 h with corresponding mass gain of 1.25%.



Fig. 4 Isothermal oxidation curves of coated samples in air at 1773 K

As shown in Fig. 4, the oxidation curve of ZrB_2 modified coating strictly obeys the exponential damping law during oxidation. At initial oxidation stage, the coating materials are exposed to air and react with oxygen directly. SiC, Si, MoSi₂ and ZrB_2 are respectively oxidized according to Eqs. (2)–(6). The oxidation of SiC, Si and ZrB_2 leads to a mass gain. The oxidation of MoSi₂ leads to a mass loss due to the volatilization of MoO₃. The mass change of the samples is determined by the mixed oxidation process of SiC, Si, ZrB_2 and MoSi₂. Because the mass gain covers the mass loss, the mass gain of coated sample reaches 1.14% after oxidation for 51 h, and SiO₂ glass is formed on the surface. With increasing oxidation time, SiO₂ grows continuously and covers the whole surface, which can provide a long-term oxidation protection for C/C composites. Then, the oxidation speed of the coated sample becomes slow. The coated sample exhibits slight mass gain compared with initial oxidation stage.

$SiC(s)+2O_2(g)=SiO_2(l)+CO_2(g)$	(2	2)
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$$2SiC(s)+3O_2(g)=2SiO_2(l)+2CO(g)$$
 (3)

$$Si(s)+O_2(g)=SiO_2(l) \tag{4}$$

$$2ZrB_{2}(s)+5O_{2}(g)=2ZrO_{2}(s)+2B_{2}O_{3}(g)$$
(5)

$$2MoSi_2(s) + 7O_2(g) = 2MoO_3(g) + 4SiO_2(l)$$
 (6)

A smooth SiO₂ glass thin film is formed on the surface of coated sample after oxidation at 1773 K for 550 h (Fig. 5(a)). The film plays an important role on sealing microcracks in the coating at high temperatures. Compared with the MoSi2-SiC-Si coating after oxidation at 1773 K for 200 h reported by LI et al [5], there are no holes or bubbles in this glass layer. The microcracks in the coating caused by the crystallization of amorphous silica and the thermal expansion coefficient mismatch of coatings and SiO₂ can self-seal during oxidation test. So, the effect of microcracks on oxidation protection can be neglected. The XRD pattern of the coating after oxidation (Fig. 5(b)) shows that the coating is composed of SiO₂, ZrSiO₄ and SiC. Figure 5(c) shows the backscattered electron image of surface. The white particles oxidized coating characterized by EDS are mainly composed of Zr, Si and O. Combined with XRD analysis, the white phase in the coating can be distinguished as ZrSiO₄, which is embed in SiO₂ glassy film in the form of granules with $1-5 \mu m$ in size. ZrSiO₄ is formed when SiO₂ reacts with ZrO₂ generated from the oxidation of ZrB₂ according to Eq. (7).

$$ZrO_2(s) + SiO_2(l) = ZrSiO_4(s)$$
(7)

The high-thermally stable $ZrSiO_4$ phase dispersed in the SiO₂ film in granular shape can improve the stability of SiO₂ film at high temperatures [23], and the formation of continuous SiO₂ glass layer with $ZrSiO_4$ dispersants decreases the diffusion rate of oxygen [24,25]. The oxidation protection of SiO₂ glass is enhanced by dispersed $ZrSiO_4$ particulars. So, the ZrB_2 modified coating with $ZrSiO_4$ -SiO₂ compound glass after oxidation exhibits better anti-oxidation performance than SiC/MoSi₂-SiC-Si coating.

The cross-section SEM image of the coated sample oxidized at 1773 K for 550 h is shown in Fig. 6. No holes or penetrating cracks can be observed in the coating, which infers that the multilayer coating remains intact and the C/C matrix is almost not oxidized after oxidation. In addition, the thickness of the multilayer coating decreases from 120 to 100 μ m which indicates that the consumption rate of coating is very slow.



Fig. 5 SEM images and XRD pattern of coated samples after oxidation in air at 1773 K for 550 h: (a) SEM image of surface; (b) XRD pattern; (c) SEM image of white phase; (d) EDS analysis of white phase



Fig. 6 SEM image of cross-section of coated sample oxidized in air at 1773 K for 550 h

3.3 Oxidation resistance of multilayer coating at 1873 and 1953 K

Figure 7 illustrates the mass loss curves of the coated C/C composites in air at 1873 and 1953 K. The coated sample exhibits mass gain first and then mass loss during oxidation test at 1873 K. The oxidation behavior can be divided into three stages. At initial oxidation stage, the coated sample exhibits mass gain because of the generation of oxides on the coating surface, which is similar to the oxidation at 1773 K. From 20 to 140 h, the mass starts to decrease. After 140 h, the mass loss increases quickly with time. The mass loss is up to 1.56% after oxidation for 200 h. From the isothermal



Fig. 7 Isothermal oxidation curves of coated samples in air at 1873 and 1953 K

oxidation curve at 1953 K, it can be seen that the sample shows mass loss quickly after oxidation for 28 h. The mass loss of coated sample is 2.44% after 50 h.

The surface and cross-section micrographs of the coating after oxidation at 1873 K for 200 h and 1953 K for 50 h are shown in Fig. 8. It can be seen that similar $ZrSiO_4$ -SiO₂ structures are formed on the surface, as shown in Figs. 8(a) and (b). And some bubbles caused by the escape of gas (like SiO₂, CO, CO₂ and B₂O₃) can be observed on the coating surface after oxidation at 1953 K (Fig. 8(b)). The SEM images of the cross-section

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Fig. 8 Surface and cross-section OM images of multilayer coating after oxidation at 1873 K for 200 h and 1953 K for 50 h: (a) Surface, 1873 K; (b) Surface, 1953 K; (c) Cross-section, 1873 K; (d) Cross-section, 1953 K

(Figs. 8(c) and (d)) show the existence of oxidation marks of C/C matrix at the end of penetrating cracks. When the coated samples are taken out of the furnace for mass, the coating suffers higher stress at 1873 and 1953 K compared with oxidation at 1773 K, which results in the formation of penetrating cracks in the coating. Oxygen can diffuse through these cracks and result in the oxidation of C/C composites. In addition, some oxidation holes in the coating can be observed after oxidation at 1953 K for 50 h as shown in Fig. 8(d), which may be attributed to the excessive pressure of oxidation gas.

 $ZrSiO_4$ starts to decompose at 1829 K, and its decomposition reaction accelerates significantly with the increase of temperature [18]. The XRD patterns of coated samples after oxidation at 1873 and 1953 K (Fig. 9) infer that the amount of $ZrSiO_4$ in coating



Fig. 9 XRD patterns of multilayer coating after oxidation at 1873 K for 200 h and 1953 K for 50 h

reduces after oxidation at 1953 K. With increasing temperature, the decomposition of $ZrSiO_4$ reduces the stability of SiO_2 glass, and also accelerates the volatilization of SiO_2 , which results in the reduction of anti-oxidation performance of coated sample at 1953 K.

4 Conclusions

1) A dense ZrB_2 modified silicon-based multilayer coating is successfully prepared on the surface of C/C composites by pack cementation. The introduction of ZrB_2 can improve the anti-oxidation performance of silicon-based coating for C/C composites.

2) The coating can effectively protect C/C composites from the oxidation at 1773 K for more than 550 h. It is attributed to the formation of stable $ZrSiO_4$ -SiO₂ complexes which effectively improves the stability of SiO₂ glassy film.

3) The mass loss is 1.56% and 2.44% after oxidation at 1873 K for 200 h and 1953 K for 50 h, respectively. With increasing temperature, the decomposition of $ZrSiO_4$ and volatilization of SiO_2 results in the decrease of anti-oxidation performance.

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炭/炭复合材料 ZrB2 改性硅基复合涂层的高温抗氧化性能

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摘 要:为提高硅基涂层的高温抗氧化性能,采用包埋法在炭/炭(C/C)复合材料表面制备 ZrB₂ 改性硅基涂层,研 究涂层的相组成、微观形貌及其在 1773、1873 和 1953 K 时的抗氧化性能。所制备的涂层结构致密,具有良好的 抗氧化性能。在 1773 K 下形成的稳定的 ZrSiO₄—SiO₂氧化保护膜对 C/C 复合材料提供 550 h 的氧化保护。涂层试 样的抗氧化性能随温度的提高而降低,在 1953 K 下氧化 50 h 其质量损失为 2.44%,这主要归因于 ZrSiO₄ 的高温 分解和 SiO₂ 保护膜随温度升高的挥发。

关键词:炭/炭复合材料;涂层;ZrB₂;抗氧化性能